

12. (Amended) Arrangement according to claim 10 [or 11], characterised in that integration routine (8) integrates out the aircraft's attitude from the aircraft's body-frame angular rates (p, q and r) obtained from the aircraft's body-frame angular rate gyros.

24. (Amended) Arrangement according to [any of claims 16 - 23] claim 16, characterised in that the first filter (11) and/or the second filter (22) consists of a Kalman filter.

Remarks

Applicants have amended the claims to eliminate multiple dependencies.

Respectfully submitted,

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